



(Ans. 8.9 m/s<sup>2</sup>; 765 N)

**Solution:** Their force of attraction = centrifugal force

Centirfugal force = 
$$\frac{mv^2}{r}$$

=

$$=\frac{86 \times \left(\frac{26040 \times 1000}{60 \times 60}\right)}{\left(\frac{12680 \times 10^{3}}{2} + 916 \times 10^{3}\right)}N$$

= 760.65 N (Weight)

Q1.5	Convert the following rea barometer reads 760 mmH	dings of pressure to kPa, assuming that the
	(a) 90 cmHg gauge	(b) 40 cmHg vacuum
	(c) 1.2 m H <sub>2</sub> O gauge	(d) 3.1 bar

Solution:  $760 \text{ mm Hg} = 0.760 \times 13600 \times 9.81 \text{ Pa}$ = 10139.16 Pa  $\approx 101.4 \text{ kPa}$ Page 7 of 265

# Introduction

By: S K Mondal

Chapter 1



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#### Temperature

# **By: S K Mondal**

Chapter 2

**Q2.4** when the reference junction of a thermocouple is kept at the ice point and the test junction is at the Celsius temperature t, and e.m.f. e of the thermocouple is given by the equation

 $\varepsilon = at + bt^2$ 

Where a = 0.20 mV/deg, and  $b = -5.0 \times 10^{-4}$  mV/deg<sup>2</sup>

- (a) Compute the e.m.f. when  $t = -100^{\circ}$ C, 200°C, 400°C, and 500°C, and draw graph of  $\varepsilon$  against t in this range.
- (b) Suppose the e.m.f.  $\varepsilon$  is taken as a thermometric property and that a temperature scale  $t^*$  is defined by the linear equation.

$$t^* = a'\varepsilon + b'$$

And that  $t^* = 0$  at the ice point and  $t^* = 100$  at the steam point. Find the numerical values of a' and b' and draw a graph of  $\varepsilon$  against  $t^*$ .

- (c) Find the values of  $t^*$  when  $t = -100^{\circ}$ C, 200°C, 400°C, and 500°C, and draw a graph of t\* against t.
- (d)
- Solution: Try please
- (d) Compare the Celsius scale with the  $t^*$  scale. Try please The temperature t on a thermometric scale is defined in terms of a Q2.5 property K by the relation

 $a = a \ln K + b$ 

Where ond b are constants. e values of K are Equina to be 1.83 and 6.78 at the ice point and the steam point the temperatures of which are assigned the numbers 0 and 100 respectively. Determine the temperature corresponding to a reading of *K* equal to 2.42 on the thermometer.

(Ans. 21.346°C)

Solution:		$t = a \ln x + b$ $0 = a x \ln 1.83 + b$ $100 = a x \ln 6.78 + b$	(i) (ii)
		Equation $\{(ii) - (i)\}$ gives	
		$\mathbf{a} \cdot \mathbf{ln} \cdot \left(\frac{6.78}{1.83}\right) = 100$	
	or	a = 76.35	
	<i>.</i> :.	$\mathbf{b} = -\mathbf{a} \times \mathbf{ln} \ 1.83$	
		= -46.143	
	:.	$t = 76.35 \ln k - 46.143$	
	<i>.</i>	$t^* = 76.35 \times \ln 2.42 - 46.143$ = 21.33°C	

Q2.6 The resistance of the windings in a certain motor is found to be 80 ohms at room temperature (25°C). When operating at full load under steady state conditions, the motor is switched off and the resistance of the windings, immediately measured again, is found to be 93 ohms. The windings are made of copper whose resistance at temperature t<sup>o</sup>C is given by

### Work and Heat Transfer

Chapter 3

$$\therefore \quad \text{Power} = \Delta p\dot{v} \\ = (0.9 - 0.101325) \times 10^{3} \text{ kPa} \times \frac{1}{60} \text{ m}^{3}/\text{s} \\ = 13.31 \text{ kJ}/\text{s} \\ \text{(b)} \quad \text{So that pressure will be 0.9 MPa} \\ \therefore \quad \text{hpg} = 0.9 \text{ MPa} \\ \text{or} \quad h = \frac{0.9 \times 10^{6}}{1000 \times 9.81} \text{ m} = 91.743 \text{ m} \\ \text{(c)} \qquad \qquad \frac{1}{2} \dot{m} \left( V_{2}^{2} - V_{1}^{2} \right) = \Delta p\dot{v} \qquad \text{where } \dot{m} = \dot{v}\rho \\ \text{or} \quad \frac{1}{2} \rho \left( V_{2}^{2} - V_{1}^{2} \right) = \Delta p \\ \text{or} \qquad V_{2}^{2} - V_{1}^{2} = 2\frac{\Delta p}{\rho} \\ \text{or} \qquad V_{2}^{2} = V_{1}^{2} + 2\frac{\Delta p}{\rho} \\ = 10^{2} + \frac{2 \times (0.9 - 0.101325) \times 10^{6}}{10003} \text{ CO} \text{ K} \\ V_{2} = 41.2 \text{ m/s} \\ \end{array}$$

Q3.2 The piston of an 6ir (ngine, of area 0004) 10, moves downwards 75 mm, drawing in 0.01 28 m<sup>3</sup> of freshar from the atmosphere. The pressure in the cylindry is aniform during the process at 80 kPa, while the atmospheric pressure is 101.3 (5 kPa, are difference being due to the flow resistance in the induction pipe and the inlet valve. Estimate the displacement work done by the air finally in the cylinder.



- Q3.3 An engine cylinder has a piston of area 0.12 m<sup>3</sup> and contains gas at a pressure of 1.5 MPa. The gas expands according to a process which is represented by a straight line on a pressure-volume diagram. The final pressure is 0.15 MPa. Calculate the work done by the gas on the piston if the stroke is 0.30 m.
- Solution: Initial pressure  $(p_1) = 1.5$  MPa Final volume  $(V_1) = 0.12m^2 \times 0.3m$

By: S K Mondal

(Ans. 29.7 kJ)

## Work and Heat Transfer

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Solution:

Solution.  
Initial volume 
$$(V_1) = \frac{\pi d^2}{4} \times L$$
  
 $= \frac{\pi \times (0.15)^2}{4} \times 0.25 \text{ m}^3$   
 $= 0.00442 \text{ m}^3$   
Initial pressure  $(p_1) = 101.325 \text{ kPa.}$   
Final volume  $(V_2) = \frac{V_1}{5} = 0.000884 \text{ m}^3$   
 $p_1V_1^{1.2} = p_2V_2^{1.2}$   
Or  $p_2 = \frac{p_1V_1^{1.2}}{V_2^{1.2}} = 699.41 \approx 700 \text{ kPa}$   
Work done/unit stroke - unit cylinder (W)  
 $= \left(\frac{1.2}{1.2-1}\right) \times [p_1V_1 - p_2V_2]$   
 $= \left(\frac{101.325 \times 0.00442 - 700 \times 0.000884}{1.2-1}\right) \times 1.2$   
Notes also be a state of 265  
PIENNE  $\left(\frac{102}{1.2-1}\right) \times [p_1V_1 - p_2V_2]$   
 $= 17.55 \text{ kW}$ 

# Q3.13 Determine the total work done by a gas system following an expansion process as shown in Figure.



(Ans. 2.253 MJ)

Solution: Area under AB =  $(0.4 - 0.2) \times 50 \times 10^5$  J =  $10^6$ W = 1 MJ



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# Second Law of Thermodynamics By: S K Mondal

Chapter 6

If this system is used as a heat pump, how many MJ of heat would be available for heating for each MJ of heat input to the engine?

(Ans. 1.8 MJ)

**Solution:** COP of the Ref. is 5 So for each MJ removed from the cold body we need work

$$=\frac{1 M J}{5} = 200 \text{ kJ}$$

For 200 kJ work output of heat engine hair  $\eta = 30\%$ 

We have to supply heat =  $\frac{200 \text{ kJ}}{0.3}$  = 666.67 kJ Now COP of H.P. = COP of Ref. + 1 = 5 + 1 = 6 Heat input to the H.E. = 1 MJ  $\therefore$  Work output (W) = 1 × 0.3 MJ = 300 kJ

That will be the input to H.P.

$$\therefore (COP)_{H,P} = \frac{Q_1}{W}$$
  
$$\therefore \qquad Q_1 = (COP)_{H,P} \times W = 6 \times 300 \text{ kJ} = 1.812 \text{ CO}$$



Q6.4 An electric storage batter, which can eachange heat only with a constant temperature a mosphere goes through a complete cycle of two processes. In process 1-2, 2.8 kWk of electrical work flow into the battery while could be flow out of the atmosphere. During process 2-1, 2.4 Wh of work flow out of the battery.

- (a) Find the he of the nsfer in process 2–1.
  - (b) If the process 1-2 has occurred as above, does the first law or the second law limit the maximum possible work of process 2-1? What is the maximum possible work?
  - (c) If the maximum possible work were obtained in process 2–1, what will be the heat transfer in the process?

(Ans. (a) -708 kJ (b) Second law,  $W_{2-1} = 9348$  kJ (c)  $Q_{2-1} = 0$ )

Solution: From the first Law of thermodynamics (a) For process 1-2  $Q_{1-2} = E_2 - E_1 + W_{1-2}$   $-732 = (E_2 - E_1) - 10080$   $[2.8 \text{ kWh} = 2.8 \times 3600 \text{ kJ}]$   $\therefore E_2 - E_1 = 9348 \text{ kJ}$ For process 2-1  $Q_{21} = E_1 - E_2 + W_{21}$  = -9348 + 8640 = -708 kJ i.e. Heat flow out to the atmosphere. (b) Yes Second Law limits the maximum possible work. As Electric end

(b) Yes Second Law limits the maximum possible work. As Electric energy stored in a battery is High grade energy so it can be completely converted to the work. Then,

# Second Law of Thermodynamics By: S K Mondal

Solution:

$$\frac{Q_1}{694} = \frac{Q_2}{T} = \frac{Q_1 - Q_2}{694 - T} = \frac{Q_3}{277.4} = \frac{Q_2 - Q_3}{T - 277.4}$$
Hence  $Q_1 - Q_2 = 2 W_2$   
 $Q_2 - Q_3 = W_2$   
 $\therefore \qquad \frac{2}{694 - T} = \frac{1}{T - 277.4}$ 
or  $2T - 277.4 \times 2 = 694 - T$   
or  $T = 416.27 \text{ K} = 143.27^{\circ} \text{ C}$ 
(b)  $\eta_1 = 40\%$ 

(c) 
$$\eta_2 = 1 - \frac{277.4}{416.27} = 33.36\%$$
  
 $Q_2 = \frac{416.27}{694} \times 200 \text{ kJ} = 119.96 \text{ kJ};$   
 $Q_1 = \frac{277.4}{416.27} \times 119.96 = 79.94 \text{ kJ}$ 

$$\begin{array}{c} 694 \text{ K} \\ \hline \mathbf{Q}_1 \\ \hline \mathbf{HE} \\ \hline \mathbf{Q}_2 \\ \hline \mathbf{$$

Chapter 6

Q6.11 A heat engine operates between the maximum and minimum temperatures of 671°C and 60°C respectively, with an efficiency of 50% of the appropriate Carnot efficiency. It drives a heat pump which uses river water at 4,4°C to heat a block of that in which the temperature is to be maintained at 21.1°C. Assuming that a temperature difference of 11.1°C exists between the working fluid and the river water, on the one that and the required room temperature on the other, and assuming the heat pump to operate on the reversed Carnot cycle, but with a COP of 50% of the ideal COP, find the heat input to the engine per unit heat output from the heat pump. Why is direct heating thermodynamically more wasteful?

(Ans. 0.79 kJ/kJ heat input)

Solution: Carnot efficiency 
$$(\eta) = 1 - \frac{273 + 60}{273 + 671} = 1 - \frac{333}{944} = 0.64725$$

Actual ( $\eta$ ) = 0.323623 = 1 -  $\frac{Q_1}{Q_1}$ 

# Second Law of Thermodynamics By: S K Mondal

Work output (W) = $\frac{Q_1}{T_1} (T_1 - T_2)$	$T_1$ $T_3$
For H.P.	$\bigstar^{\mathbf{Q}_1}  \bigstar^{\mathbf{Q}_3}$
Work input (W) = $\frac{Q_4}{T_4} (T_3 - T_4)$	$(H.E) \rightarrow (W) \rightarrow (H.P)$
$\therefore  \frac{Q_1}{T_1} (T_1 - T_2) = \frac{Q_4}{T_4} (T_3 - T_4)$	$\mathbf{P}_{\mathbf{Q}_{2}}$ $\mathbf{P}_{\mathbf{Q}_{4}}$ $\mathbf{T}_{4}$
or $\left[ \frac{\mathbf{Q}_4}{\mathbf{Q}_1} = \frac{\mathbf{T}_4}{\mathbf{T}_1} \left\{ \frac{\mathbf{T}_1 - \mathbf{T}_2}{\mathbf{T}_3 - \mathbf{T}_4} \right\} \right]$	
This is the expression.	
Prove that the following propositions a (a) A PMM2 is Impossible	are logically equivalent:

Chapter 6

- (b) A weight sliding at constant velocity down a frictional inclined plane
- Applying First Law of Thermodynamics  $Q_{12} = E_2 E_1 + W_{1,2}$ or  $0 = E_2 = E_1$  on  $P_1$   $Q_1 = E_2 = E_1$  on  $P_1$   $Q_1 = E_2 = E_1$  on  $P_1$   $Q_2 = E_2 = E_1$  on  $P_1$   $Q_1 = E_2 = E_1$  on  $P_1$   $Q_2 = E_2 = E_1$  on  $P_2$   $Q_2 = E_2 = E_1$  on  $P_1$   $Q_2 = E_2 = E_1$  on  $P_2$   $Q_3 = E_2$  on  $P_1$   $Q_3 = E_2$  on  $P_2$   $Q_3 = E_2$  on  $P_1$   $Q_3 = E_2$  on  $P_2$   $Q_3 = E_2$  on  $P_1$   $Q_3 = E_2$  on  $P_2$   $Q_3 = E_2$  on  $P_1$   $Q_3 = E_2$  on  $P_2$   $Q_3 = E_2$  on  $P_1$   $Q_3 = E_2$  on  $P_2$   $Q_3 = E_2$  on  $P_2$  Q**Solution:** Previe

**Q6.24** 



- In a Carnot cy 01, beat is supplied at 350°C and rejected at 27°C. The working fluid is water which, while receiving heat, evaporates from liquid at 350°C to steam at 350°C. The associated entropy change is 1.44 kJ/kg K.
  - (a) If the cycle operates on a stationary mass of 1 kg of water, how much is the work done per cycle, and how much is the heat supplied?
  - (b) If the cycle operates in steady flow with a power output of 20 kW, what is the steam flow rate?

(Ans. (a) 465.12, 897.12 kJ/kg, (b) 0.043 kg/s)

**Solution:** If heat required for evaporation is Q kJ/kg then

(a) 
$$\frac{Q}{(350 + 273)} = 1.44$$
  
or Q = 897.12 kJ/kg  
It is a Carnot cycle so  $\eta = 1 - \frac{(273 + 27)}{(350 + 273)}$   
 $\therefore$  W =  $\eta$ .Q = 465.12 kJ  
(b) P =  $\dot{m}W$  or  $\dot{m} = \frac{P}{W} = \frac{20}{465.12}$  kg/s = 0.043 kg/s

Q7.5 A heat engine receives reversibly 420 kJ/cycle of heat from a source at  $327^{\circ}$ C, and rejects heat reversibly to a sink at  $27^{\circ}$ C. There are no other heat transfers. For each of the three hypothetical amounts of heat rejected, in (a), (b), and (c) below, compute the cyclic integral of  $\frac{dQ}{T}$ . Page 72 of 265



Q7.8 One kg of air initially at 0.7 MPa, 20°C changes to 0.35 MPa, 60°C by the three reversible non-flow processes, as shown in Figure. Process 1: *a*-2 consists of a constant pressure expansion followed by a constant volume cooling, process 1: *b*-2 an isothermal expansion followed by a constant pressure expansion, and process 1: *c*-2 an adiabatic

### Entropy

By: S K Mondal

**Chapter 7** 

$$S_4 - S_3 = \int_{273}^{268} \frac{m c_p dT}{T} = 0.01 \times \left(\frac{4.2}{2}\right) \times \ln \frac{268}{273} \text{ kJ/ } K$$
$$= -0.3882 \text{ J/K}$$

: 
$$S_4 - S_1 = -15.63 \text{ J/K}$$

Net Entropy change = 15.63 J/K*.*..

Q7.10

Calculate the entropy change of the universe as a result of the following processes:

- (a) A copper block of 600 g mass and with  $C_p$  of 150 J/K at 100°C is placed in a lake at 8°C.
- -, are joined by other. (Area: 5.69 J/K, (b) 2.095 J/K, (c) 3.64 J/K) 000 06 265 (b) The same block, at 8°C, is dropped from a height of 101 h into the lake.
- (c) Two such blocks, at 100 and 0°C, are joined ogether.

#### Solution:

(a)

100 m

42.4<mark>8</mark> J/K As unit of  $C_P$  is J/K there for *.*.. It is heat capacity

i.e. 
$$C_p = m c_p$$

 $(\Delta S)_{copper} =$ 

$$(\Delta S)_{lake} = \frac{C_p (100 - 8)}{281} J/K$$
$$= \frac{150(100 - 8)}{281} J/K = 49.11 J/K$$
$$(\Delta S)_{univ} = (\Delta S)_{COP} + (\Delta S)_{lake} = 6.63 J/K$$

(b) Work when it touch water =  $0.600 \times 9.81 \times 100 \text{ J} = 588.6 \text{ J}$ As work dissipated from the copper

$$(\Delta S)_{copper} = 0$$
As the work is converted to heat and absorbed by water then  

$$(\Delta S)_{lake} = \frac{W = Q}{281} = \frac{588.6}{281} J/K = 2.09466 J/K$$
∴ (ΔS) univ = 0 + 2.09466 J/k = 2.09466 J/K  
(c) Final temperature (T<sub>f</sub>) =  $\frac{100 + 0}{2} = 50^{\circ} C = 323 K$ 

### Entropy

# By: S K Mondal

Chapter 7

(ii) the change in entropy for the room air, (iii) the work done by the engine.

(b) If the aluminium block is allowed to cool by natural convection to room air, compute (i) the change in entropy for the block, (ii) the change in entropy for the room air (iii) the net the change in entropy for the universe.

> (Ans. (a) – 134 J/K, + 134 J/K, 740 J; (b) – 134 J/K, + 136.5 J/K, 2.5 J/K)

#### Solution:

(a)

$$(\Delta S)_{A1} = \int_{313}^{293} \frac{m c_p dT}{T}$$

$$5 \times 400 \times \ln \frac{293}{313} J/K = -132.06 J/K$$

$$(\Delta S)_{air} = \frac{Q - W}{293}$$
And Q = m c<sub>p</sub> (313 - 293) = 40000 J  
As heat is reversibly flow then  

$$(\Delta S)_{A1} + (\Delta S)_{air} = 0$$
or  $-132.06 + 136.52 - \frac{W}{392} = 90$  Sale C  $V$   
or  $V = 1.806 \text{ kJ}$  265  
(b) (A5) when for reversible or pre-ersible = -132.06 J/K
$$(\Delta S)_{air} = \frac{4000}{253} 218055 J/K$$

$$(\Delta S)_{air} = +4.4587 J/K$$

Q7.22 Two bodies of equal heat capacities C and temperatures  $T_1$  and  $T_2$  form an adiabatically closed system. What will the final temperature be if one lets this system come to equilibrium (a) freely? (b) Reversibly? (c) What is the maximum work which can be obtained from this system?

Solution:

Freely 
$$T_f = \frac{T_1 + T_2}{2}$$

(b) Reversible  
Let find temperature be 
$$T_f$$
  
the  $(\Delta S)_{hot} = \int_{T_1}^{T_f} C \frac{dT}{T}$   
 $= C \ln \frac{T_f}{T_1}$   
 $(\Delta S)_{cold} = \int_{T_2}^{T_f} C \frac{dT}{T} = C \ln \left(\frac{T_f}{T_2}\right)$   
 $\therefore (\Delta S)_{univ.} = (\Delta S)_{hot} = (\Delta S)_{cold} = (\Delta S)_{surroundings}$   
 $= C \ln \frac{T_f}{T_1} + C \ln \frac{T_f}{T_2} = 0$   
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### Entropy

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2 kg

Solution:

$$(\Delta S)_{surr.} = 0$$

$$(\Delta S)_{\rm sys} = \int_{298}^{303} {\rm m} \, c \, \frac{{\rm d} T}{T}$$

$$= 2 \times 4.187 \times \ln \frac{303}{298} = 0.13934 \text{ kJ/K}$$
  

$$\therefore \quad (\Delta S)_{univ} = (\Delta S)_{sys} + (\Delta S)_{surr} = 0.13934 + 0 = 0.13934 \text{ kJ/K}$$

Q7.25 A copper rod is of length 1 m and diameter 0.01 m. One end of the rod is at 100°C, and the other at 0°C. The rod is perfectly insulated along its length and the thermal conductivity of copper is 380 W/mK. Calculate the rate of heat transfer along the rod and the rate of entropy production due to irreversibility of this heat transfer.

$$\dot{Q} = kA \frac{\Delta T}{\Delta x}$$
  
= 380 × 7.854 × 10<sup>-5</sup> ×  $\frac{100}{1}$  W = 2.9845 W

At the 373 K end from surrounding  $\hat{\mathbf{Q}}$  amount heat is go to the system. So at this end

$$(\Delta \dot{S})_{charge} = -\frac{\dot{Q}}{373}$$

And at the 273 K and from system  $\hat{Q}$  amount of heat is rejected to the surroundings.

$$\therefore \qquad (\Delta \dot{S})_{charge} = \frac{\dot{Q}}{273}$$
$$\therefore \qquad (\Delta \dot{S})_{univ.} = \frac{\dot{Q}}{273} - \frac{\dot{Q}}{373} = 0.00293 \text{ W/K}$$

Q7.26 A body of constant heat capacity  $C_p$  and at a temperature  $T_i$  is put in contact with a reservoir at a higher temperature  $T_f$ . The pressure remains constant while the body comes to equilibrium with the reservoir. Show that the entropy change of the universe is equal to

# Entropy By: S K Mondal

Solution: Try please.

Q7.31 A block of iron weighing 100 kg and having a temperature of 100°C is immersed in 50 kg of water at a temperature of 20°C. What will be the change of entropy of the combined system of iron and water? Specific heats of iron and water are 0.45 and 4.18 kJ/kg K respectively.

(Ans. 1.1328 kJ/K)

Solution: Let final temperature is tf °C *.*..  $100 \times 0.45 \times (100 - t_f) = 50 \times 4.18 \times (t_f - 20)$  $100 - t_f = 4.644 t_f - 20 \times 4.699$  $5.644 t_{\rm f} = 192.88$ or  $t_f = 34.1732^{\circ} C$ or  $t_f = 307.1732 \text{ K}$ *.*.. ENTROPY =  $(\Delta S)_{iron} + (\Delta S)_{water}$  $= 100 \times 0.45 \ln\left(\frac{307.1732}{373}\right) + 50 \times 4.180 \times \ln\left(\frac{307.1732}{293}\right)$ = 1.1355 kJ/K 36 g of water at 30°C are converted into steam at 70°C Q7.32 at constant atmospheric pressure. The specific heat of with is assumed constant at 4.2 J/g K and the latent heat of vaporization at 100°C is 2260 J/g. For water vapour, assume pV = mRT (1) r = 0.4619 kJ/kg K, and  $\frac{C_p}{R} = a + bT + cT$ hange of the system. (Ans. 277.8 J/K) Solution: m = 36 g = 0.036 kg $T_1 = 30^{\circ}C = 303 \text{ K}$  $T_2 = 373 \text{ K}$  $T_3 = 523 \text{ K}$ 

$$(\Delta S)$$
 Water

$$= m c_{p} \ln \left(\frac{373}{303}\right) kJ/K$$
  
= 0.03143 kJ/K  
( $\Delta S$ ) Vaporization =  $\frac{mL}{T_{2}}$   
=  $\frac{0.036 \times 2260}{373}$   
= 0.21812 kJ/K  
( $\Delta S$ ) Vapor =  $\int_{373}^{523} m c_{p} \frac{dT}{T}$   
= mR  $\int_{373}^{523} (\frac{a}{T} + b + CT) dT$ 



#### Availability & Irreversibility

# By: S K Mondal

**Q8.8** 

Chapter 8

:. Required mass of air =  $\frac{5200}{363.36}$  kg = 14.311 kg

Specific volume of air at 7 MPa, 25°C then

$$v = \frac{RT}{p} = \frac{0.287 \times 298}{7000} m^3/kg = 0.012218 m^3/kg$$

 $\therefore$  Required storage volume (V) = 0.17485 m<sup>3</sup>

Ice is to be made from water supplied at  $15^{\circ}$ C by the process shown in Figure. The final temperature of the ice is –  $10^{\circ}$ C, and the final temperature of the water that is used as cooling water in the condenser



(ii)  $W_R$  required for  $0^\circ$  C water to  $0^\circ$  C ice

#### Availability & Irreversibility

By: S K Mondal

 $W_{act} = \frac{181.55}{0.75} = 242 \text{ kJ/kg}$ (a) ... Power required driving the compressor  $= \dot{m} W_{act} = 2.42 \text{ kW}$ Extra work addede in 2' to 2 is (242 - 181.55) = 60.85 kJ/kgIf  $C_p(T_2 - T_2) = 60.85$ ....  $T_2 = T'_2 + \frac{60.85}{1.005} = 529.25 \text{ K}$ or *.*.. Availability loss due to cooling  $= \int_{0}^{0.029,250} 1 \times 1.005 \left( 1 - \frac{288}{T} \right) dT$  $= 1.005 \left\{ (529.21 - 313) - 288 \ln \left( \frac{529.25}{313} \right) \right\} \text{ kJ/kg}$ = 65.302 kJ/kgTotal available energy loss *.*.. = (60.85 + 65.302) kJ/kg = 23(1) k Power loss due to irreversibility *.*.. In a rotary compressor, producers at 11 bar, 21 ° C where it is compressed adiabatically to 6.6 bar 250 Conculate the irreversibility Q8.13 and the entropy production for untribass flow rate. The atmosphere is at 1.03 b 2.10 C. Neglect the L.C. changes. (Ans. 19 kJ/kg, 0.064 kJ/kg K) Solution  $p_1 = 1.1 \text{ bar} = 110 \text{ kPa}$  $T_1 = 294 \text{ K}$  $p_2 = 6.6 \text{ bar} = 660 \text{ kPa}$  $T_2 = 523 \text{ K}$  $p_0 = 103 \text{ kPa}$ Т  $T_0 = 293 \text{ K}$  $\Delta \mathbf{s} = \mathbf{s}_2 - \mathbf{s}_1 = \int_{-\infty}^{2} \left( \frac{d\mathbf{h}}{T} - \mathbf{v} \frac{dp}{T} \right)$  $\mathbf{S}$  $= \left| \operatorname{C}_{\mathrm{P}} \ln \frac{\mathrm{T}_{2}}{\mathrm{T}_{1}} - \operatorname{R} \ln \frac{p_{2}}{p_{1}} \right|$  $= \left| 1.005 \ln \frac{523}{294} - 0.287 \ln \left( \frac{660}{110} \right) \right|$ = 0.064647 kJ/kg - K = 64.647 J/kg - KMinimum work required  $W_{min} = Availability$  increase  $= h_2 - h_1 - T_0 (s_2 - s_1)$  $= mc_{P} (T_{2} - T_{1}) - T_{0}\Delta s$ 

=  $1 \times 1.005 (523 - 294) - 293 \times 0.064647 = 211.2 \text{ kJ/kg}$ Page 107 of 265

### Availability & Irreversibility

the plant per kg of turbine gas? Take  $c_p$  for exhaust gas as 1.08 and

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Chapter 8

for air as 1.05 kJ/kg K. Neglect heat transfer to the surroundings and the changes in kinetic and potential energy. (Ans. (a) 66 kJ/kg, (b) 0.0731 kJ/kg K, (c) 38.7 kJ/kg) Availability decrease of extra gases =  $Q\left(1 - \frac{T_0}{T}\right)$ Solution: (a)  $= \int_{700}^{800} \mathrm{m} c_p \left( 1 - \frac{293}{\mathrm{T}} \right) \mathrm{dT} = 1 \times 1.08 \left[ (800 - 700) - 293 \ln \left( \frac{800}{700} \right) \right]$ = 65.745 kJ/kgGas Т tesale.co.uk 01  $\times c_{pg} (800-700)$ m  $T_{e} = 521.5 \text{ K}$ or Availability increases *.*.. =  $2 \times 1.05 \times \left[ (521.5 - 470) - 293 \ln \frac{521.5}{470} \right] = 44.257 \text{ kJ/kg}$  $\therefore$  S<sub>gas</sub> = 73.336 J/K of per kg gas flow For reversible heat transfer  $(\Delta S)_{univ} = 0$  $(\Delta S)_{Gas} = -(\Delta S)_{water}$  $m \times 1.08 \ln \frac{800}{700}$  $= -2 \,\mathrm{m} \times 1.05 \times \ln\left(\frac{470}{\mathrm{T}}\right)$  $\ln \frac{T_{o}}{470} = 0.068673$ or  $T_0 = 503.4 \text{ K}$  $Q_1 = m \times 1.08(800 - 700) = 108 \text{ kJ/kg}$  $Q_2 = 2m \times 1.05 (503.4 - 470) = 70.162 \text{ kJ/kg of gas}$  $W = Q_1 - Q_2 = 37.84 \text{ kJ/kg of gas flow}$ [i.e. extra output]

Q8.18 An air preheater is used to heat up the air used for combustion by cooling the outgoing products of combustion from a furnace. The rate of flow of the products is 10 kg/s, and the products are cooled from 300°C to Page 110 of 265

# **Properties of Pure Substances**

# By: S K Mondal

Chapter 9

(Ans. 0.944, 0.921)

$$=\frac{\mathbf{x}_2 \times 4.2}{4.2 + 0.55} = \frac{0.96216 \times 4.2}{4.2 + 0.55} = 0.85$$

 ${\bf h}_{\rm 1}{=}~{\bf h}_{\rm f1}{}+{\bf x}~{\bf h}_{\rm fg1}{}=844.7{}+0.85{}\times{}1945.2{}=2499.6{}\;{\rm kJ/kg}$ 

But at 1 bar minimum 5° super heat i.e. 105°C enthalpy is 2686 kJ/kg So it is not possible to calculate only by throttling calorimeter.

Q9.11 Steam from an engine exhaust at 1.25 bar flows steadily through an electric calorimeter and comes out at 1 bar, 130°C. The calorimeter has two 1 kW heaters and the flow is measured to be 3.4 kg in 5 min. Find the quality in the engine exhaust. For the same mass flow and pressures, what is the maximum moisture that can be determined if the outlet temperature is at least 105°C?

Solution: 
$$h_2 = 2676.2 + \frac{30}{50} (2776.4 - 2676.2) = 2736.3 \text{ kJ/kg}$$
  
 $\dot{\mathbf{m}} h_1 = \dot{\mathbf{m}} h_2 - \dot{\mathbf{Q}}$   
 $\mathbf{m} = \frac{3.4 \text{ kg}}{1.25 \text{ bar}}$ 
 $\mathbf{m} = \frac{3.4 \text{ kg}}{1.30^{\circ} \text{ c}}$ 
 $\mathbf{m} = \frac{3.24 \text{ kg}}{1.30^{\circ} \text{ c}}$ 
 $\mathbf{m} = \frac{3.23 \text{ kg}}{1.30^{\circ} \text{ c}}$ 
 $\mathbf{m} = \frac{3.23 \text{ kg}}{1.30^{\circ} \text{ c}}$ 
 $\mathbf{m} = \frac{$ 

12 Steam expands isentropically in a nozzle from 1 MPa, 250°C to 10 kPa. The steam flow rate is 1 kg/s. Find the velocity of steam at the exit from the nozzle, and the exit area of the nozzle. Neglect the velocity of steam at the inlet to the nozzle.

### **Properties of Pure Substances**

# By: S K Mondal

**Chapter 9** 

Q9.18

A sample of wet steam from a steam main flows steadily through a partially open valve into a pipeline in which is fitted an electric coil. The valve and the pipeline are well insulated. The steam mass flow rate is 0.008 kg/s while the coil takes 3.91 amperes at 230 volts. The main pressure is 4 bar, and the pressure and temperature of the steam downstream of the coil are 2 bar and 160°C respectively. Steam velocities may be assumed to be negligible.

- (a) Evaluate the quality of steam in the main.
- (b) State, with reasons, whether an insulted throttling calorimeter could be used for this test.

(Ans. (a) 0.97, (b) Not suitable)

Solution: (a)



From steady flow energy equation

$$\dot{\mathbf{m}} \mathbf{h}_2 + \dot{\mathbf{Q}} = \dot{\mathbf{m}} \mathbf{h}_3 + 0 : \mathbf{h}_2 = \mathbf{h}_3 - \frac{\dot{\mathbf{Q}}}{\dot{\mathbf{m}}} = 2676.73 \text{ kJ/kg}$$

If dryness fraction of steam x then

$$h_2 = h_{f2} + x h_{fg2}$$
  
or 2676.73 = 604.7 + x × 2133  $\therefore$  x= 0.9714

- (b) For throttling minimum enthalpy required 2686 kJ/kg if after throttling 5°C super heat and atm. Pressure is maintained as here enthalpy is less so it is not possible in **throttling calorimeter**.
- Q9.19 Two insulated tanks, A and B, are connected by a valve. Tank A has a volume of 0.70 m<sup>3</sup> and contains steam at 1.5 bar, 200°C. Tank B has a volume of 0.35 m<sup>3</sup> and contains steam at 6 bar with a quality of 90%. The valve is then opened, and the two tanks come to a uniform state. If there is no heat transfer during the process, what is the final pressure? Compute the entropy change of the universe.

		( <b>Ans.</b> 322.6 KPa, 0.1985 kJ/K)
Solution:	From Steam Table	from Steam Table
	Sp. Enthalpy $(h_A) = 2872.9 \text{ km}$	J/kg $t_{\rm B} = 158.8^{\circ} {\rm C}$
	Sp. Vol $(v_A) = 1.193 \text{ m}^3/\text{kg}$	Page 139 of 265 $(h_B)$

#### **Properties of Gases and Gas Mixtures By: S K Mondal** Chapter 10

a =  $3 p_c v_c^2$ ; b =  $\frac{v_c}{3}$ ; R =  $\frac{8}{3} \frac{p_c v_c}{T_c}$ ; values of Z at critical **point 0.375 for Van der** Waal 10. gas.

**Boyle temperature**  $(T_B) = \frac{a}{bB}$ 11.

$$12. \qquad \mu = x_1 \mu_1 + x_2 \mu_2 + \dots + x_c \mu_c \\ R_m = \frac{m_1 R_1 + m_2 R_2 + \dots + m_c R_c}{m_1 + m_2 + \dots + m_c R_c} \\ u_m = \frac{m_1 u_1 + m_2 u_2 + \dots + m_c u_c}{m_1 + m_2 + \dots + m_c}; \qquad h_m = \frac{m_1 h_1 + m_2 h_2 + \dots + m_c h_c}{m_1 + m_2 + \dots + m_c h_c} \\ C_{pm} = \frac{m_1 c_{p_1} + m_2 c_{p_2} + \dots + m_c c_{p_c}}{m_1 + m_2 + \dots + m_c R_c} R_c \ln \frac{p_c}{p} \end{bmatrix}$$

**Gibbs function** G =  $\overline{R}T \sum n_x (\phi_k + \ln p + \ln x_k)$ 13.

unction 
$$G = \overline{R}T \sum n_x (\phi_k + \ln p + \ln x_k)$$
  
Questions with Solution **F**K. Nag

Q.10.1 What is the mass of air contained in a room  $6 \text{ m} \times 9 \text{ m} \times 4 \text{ m}$  if the resture is 101.325 line and the temperature is 25°C? (Ans. 256 kg) Given pressure (p) = 101.325 kPa Solution:  $= 25^{\circ}C = (25 + 273) \text{ K} = 298 \text{ K}$ Temperature (T)

 $= 6 \times 9 \times 4 \text{ m}^3 = 216 \text{ m}^3$ Volume (V) From equation of states pV = mRTFor air R = 0.287 kJ/kg - K, Gas constant mass is m kg

$$m = \frac{pV}{RT} = \frac{101.325 \times 216}{0.287 \times 298} \text{ kg} = 255.9 \text{ kg}$$

- Q.10.2 The usual cooking gas (mostly methane) cylinder is about 25 cm in diameter and 80 cm in height. It is changed to 12 MPa at room temperature (27°C).
  - (a) Assuming the ideal gas law, find the mass of gas filled in the cylinder.
  - (b) Explain how the actual cylinder contains nearly 15 kg of gas.
  - (c) If the cylinder is to be protected against excessive pressure by means of a fusible plug, at what temperature should the plug melt to limit the maximum pressure to 15 MPa?
- (D) = 25 cm = 0.25 mSolution: Given diameter (H) = 80 cm = 0.8 mHeight  $(V) = \frac{\pi D^2}{4} \times H = 0.03927 \text{ m}^3$ Volume of the cylinder .... Page (p) = 12 MPa = 12000 kPaGas pressure

## **Properties of Gases and Gas Mixtures**



If the containers are connected, allowing the gases to mix freely, find (a) the pressure and temperature of the resulting mixture at equilibrium, and (b) the change of entropy of each constituent and that of the mixture.

**Solution :** Try please.

- Q.10.27 Calculate the volume of 2.5 kg moles of steam at 236.4 atm. and 776.76 K with the help of compressibility factor versus reduced pressure graph. At this volume and the given pressure, what would the temperature be in K, if steam behaved like a van der Waals gas? The critical pressure, volume, and temperature of steam are 218.2 atm., 57 cm<sup>3</sup>/g mole, and 647.3 K respectively.
- **Solution :** Try please.
- Q.10.28 Two vessels, A and B, each of volume 3 m<sup>3</sup> may be connected together by a tube of negligible volume. Vessel a contains air at 7 bar, 95°C while B

# Properties of Gases and Gas Mixtures By: S K Mondal Chapter 10

its original state. Draw the cycle on p-v and T'-s coordinates. Show that the cycle efficiency can be expressed in the following form

$$\eta = 1 - \frac{(\gamma - 1) \ln r}{\gamma \left[ r^{\gamma - 1/\gamma} - 1 \right]}$$

Where r is the pressure ratio,  $p_2/p_1$ . Determine the pressure ratio and the cycle efficiency if the initial temperature is 27°C and the maximum temperature is 327°C.

(Ans. 13.4, 32.4%)

**Solution:** Heat addition  $(Q_1) = Q_{2-3} = mc_n(T_3 - T_2)$ 



 $\begin{array}{l} \Rightarrow & \text{ If initial temperature } (T_1) = 27^\circ \text{C} = 300 \text{ K} = T_2 \\ \text{And} & T_3 = 327^\circ \text{C} = 600 \text{ K} \\ \hline & \ddots & r = \left(\frac{T_3}{T_1}\right)^{\frac{\gamma}{\gamma-1}} = \left(\frac{600}{300}\right)^{\frac{1.4}{1.4-1}} = 11.314 \\ \hline & \ddots & \eta = 1 - \frac{(1.4-1)}{(1.4)} \times \frac{\ln(11.314)}{\left[ (11.314)^{\frac{1.4-1}{1.4}} - 1 \right]} = 0.30686 \end{array}$ 

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# **Properties of Gases and Gas Mixtures**

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Q10.49 The relation between u, p and v for many gases is of the form u = a + bpvwhere a and b are constants. Show that for a reversible adiabatic process  $pv^{y} = constant$ , where

$$\gamma = (\mathbf{b} + 1)/\mathbf{b}.$$

Solution: Try please.

Q10.50 (a) Show that the slope of a reversible adiabatic process on p-v coordinates is

$$\frac{d\mathbf{p}}{d\mathbf{v}} = \frac{1}{\mathbf{kv}} \frac{\mathbf{c}_{p}}{\mathbf{c}_{v}} \text{ where } \mathbf{k} = -\frac{1}{\mathbf{v}} \left( \frac{\partial \mathbf{v}}{\partial \mathbf{p}} \right)_{\mathrm{T}}$$

(b) Hence, show that for an ideal gas,  $pv^{\gamma} = constant$ , for a reversible adiabatic process.

Solution: Try please.

Q10.51 A certain gas obeys the Clausius equation of state p(v - b) = RT and has its internal energy given by  $u = c_v T$ . Show that the equation for a reversible adiabatic process is  $p(v-b)^{\gamma} = \text{constant}$ , where  $\gamma = c_p / c_v$ .

**Solution:** Try please.

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Again

$$dx = \left(\frac{\partial x}{\partial f}\right)_{e} df + \left(\frac{\partial x}{\partial z}\right)_{f} dz$$
$$\left(\frac{\partial x}{\partial z}\right)_{f} = \left(\frac{\partial x}{\partial y}\right)_{f} \left(\frac{\partial y}{\partial z}\right)_{f}$$
$$\left(\frac{\partial x}{\partial y}\right)_{f} \left(\frac{\partial y}{\partial z}\right)_{f} \left(\frac{\partial z}{\partial x}\right)_{f} = 1$$

**Theorem 3**. Among the variables x, y, and z any one variable may be considered as a function of the other two. Thus

Among the thermodynamic variables p, V and T. The following relation holds good

$$\left(\frac{\partial p}{\partial V}\right)_T \left(\frac{\partial V}{\partial T}\right)_p \left(\frac{\partial T}{\partial p}\right)_v = -1$$

### **Maxwell's Equations**

A pure substance existing in a single phase has only two independent variables. Of the eight quantities p, V, T, S, U, H, F (Helmholtz function), and G (Gibbs function) any one may be expressed as a function of any two others.

For a pure substance undergoing an infinitesimal reversible process

(a) dU = TdS - pdV

(b) dH = dU + pdV + VdP = TdS + Vdp

(c) dF = dU - TdS - SdT = -pdT - SdT

(d) dG = dH - TdS - SdT = Vdp - SdT

Since U, H, F and G are thermodynamic properties and exact differentials of the type

dz = M dx + N dy, then

$$\left(\frac{\partial M}{\partial y}\right)_{x} = \left(\frac{\partial N}{\partial x}\right)_{y}$$

Applying this to the four equations

Chapter 11

Then 
$$S = S(T, p)$$
  
or  $dS = \left(\frac{\partial S}{\partial T}\right)_p dT + \left(\frac{\partial S}{\partial p}\right)_T dp$   
multiplying both side by T  
 $TdS = T\left(\frac{\partial S}{\partial T}\right)_p dT + T\left(\frac{\partial S}{\partial p}\right)_T dp$   
Since  $T\left(\frac{\partial S}{\partial T}\right)_p = C_p$ , heat capacity at constant pressure  
and  $\left(\frac{\partial S}{\partial p}\right)_T = -\left(\frac{\partial V}{\partial T}\right)_p$  by Maxwell's equation  
 $\therefore$   $TdS = C_p dT - T\left(\frac{\partial V}{\partial T}\right)_p dp$  proved.  
Derive:

# (iii)Derive:

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$$TdS = C_{v}dT + T\frac{\beta}{k}dV = C_{p}dT - TV\beta dp = \frac{k}{\beta} \frac{C}{\beta} \frac{dQ}{\beta} + \frac{O_{p}}{\beta V} dV$$
[IES-2001]  
We know that volume toppansivity  $(\beta) = \frac{Q}{2} \frac{Q}{2}$   
and isothermal compressibility  $(k) = -\frac{1}{V} \left(\frac{\partial V}{\partial p}\right)_{T}$ 

$$\begin{array}{ll} & \ddots & \mbox{From first TdS equation} \\ TdS = C_v dT + T \left( \frac{\partial p}{\partial T} \right)_v dV \\ \\ & \frac{\beta}{k} = - \frac{\left( \frac{\partial V}{\partial T} \right)_p}{\left( \frac{\partial V}{\partial p} \right)_T} = - \left( \frac{\partial V}{\partial T} \right)_p \cdot \left( \frac{\partial p}{\partial V} \right)_T \\ \\ As & \left( \frac{\partial V}{\partial T} \right)_p \cdot \left( \frac{\partial T}{\partial p} \right)_V \cdot \left( \frac{\partial p}{\partial V} \right)_T = -1 \\ \\ & \ddots & - \left( \frac{\partial V}{\partial T} \right)_p \cdot \left( \frac{\partial p}{\partial V} \right)_T = \left( \frac{\partial p}{\partial T} \right)_V \\ \\ or & \frac{\beta}{k} = \left( \frac{\partial p}{\partial T} \right)_V \\ \\ & \ddots & TdS = C_v dT + T \cdot \frac{\beta}{k} \cdot dV \end{array}$$

From second TdS relation

# By: S K Mondal Chapter 11 Joule-Kelvin Effect or Joule-Thomson coefficient

The value of the specific heat  $c_p$  can be determined from p-v-T data and the Joule–Thomson coefficient. The **Joule–Thomson coefficient**  $\mu_{J}$  is defined as

$$\mu_{\rm J} = \left(\frac{\partial T}{\partial p}\right)_h$$

Like other partial differential coefficients introduced in this section, the Joule–Thomson coefficient is defined in terms of thermodynamic properties only and thus is itself a property. The units of  $\mu_{J}$  are those of temperature divided by pressure.

A relationship between the specific heat  $c_p$  and the Joule–Thomson coefficient  $\mu_j$  can be established to write

$$\left(\frac{\partial T}{\partial p}\right)_{h}\left(\frac{\partial p}{\partial h}\right)_{T}\left(\frac{\partial h}{\partial T}\right)_{p} = -1$$

The first factor in this expression is the Jozle– homson coefficient fair the third is  $c_p$ . Thus

$$c_{p} = \frac{c_{p}}{\mu_{f}} (\frac{\partial p}{\partial h})_{T}$$

$$(\frac{\partial h}{\partial p})_{T} = 1/(\frac{\partial p}{\partial h})_{T}$$
this can be written as

With

$$c_{p} = -rac{1}{\mu_{J}} \left( rac{\partial h}{\partial p} 
ight)_{T}$$

The partial derivative  $(\partial h / \partial p)_T$ , called the *constant-temperature coefficient*, can be eliminated. The following expression results:

$$c_{\rm p} = \frac{1}{\mu_J} \left[ T \left( \frac{\partial v}{\partial T} \right)_p - v \right]$$

allows the value of  $c_p$  at a state to be determined using p-v-T data and the value of the Joule– Thomson coefficient at that state. Let us consider next how the Joule–Thomson coefficient can be found experimentally.

The numerical value of the slope of an isenthalpic on a *T*-*p* diagram at any point is called the *Joule-Kelvin coefficient* and is denoted by  $\mu_j$ . Thus the locus of all points at which  $\mu_j$  is zero is the inversion curve. The region inside the inversion curve where  $\mu_j$  is positive is called the *cooling region* and the region outside where  $\mu_j$  is negative is called the *heating region*. So,



Isenthalpic Curves and the Inversion Curve

#### **Energy Equation**

For a system undergoing an infinitesimal reversible process between two equilibrium states, the change of internal energy is

dU = TdS - pdV  
Substituting the first TdS equation  

$$dU = C_v dT + T \left( \frac{\partial p}{\partial T} \right) U = 0$$

$$dV = C_v dT + \left[ T \left( \frac{\partial p}{\partial T} \right) \right] dV = 0$$

$$dV = \left( \frac{\partial U}{\partial T} \right)_V dT + \left( \frac{\partial U}{\partial V} \right)_T dV$$

$$\left( \frac{\partial U}{\partial V} \right)_T = T \left( \frac{\partial p}{\partial T} \right)_V - p$$

This is known as energy equation. Two application of the equation are given below-(a) For an ideal gas,  $p = \frac{n\overline{RT}}{V}$ 

$$\therefore \left(\frac{\partial p}{\partial T}\right)_{V} = \frac{n\overline{R}}{V} = \frac{p}{T}$$
$$\therefore \left(\frac{\partial U}{\partial V}\right)_{T} = T \cdot \frac{p}{T} - p = 0$$

U does not change when V changes at T = C.

$$\begin{pmatrix} \frac{\partial U}{\partial p} \end{pmatrix}_{T} \begin{pmatrix} \frac{\partial p}{\partial V} \end{pmatrix}_{T} \begin{pmatrix} \frac{\partial V}{\partial U} \end{pmatrix}_{T} = 1$$

$$\begin{pmatrix} \frac{\partial U}{\partial p} \end{pmatrix}_{T} \begin{pmatrix} \frac{\partial p}{\partial V} \end{pmatrix}_{T} = \begin{pmatrix} \frac{\partial U}{\partial V} \end{pmatrix}_{T} = 0$$
since  $\begin{pmatrix} \frac{\partial p}{\partial V} \end{pmatrix}_{T} \neq 0, \begin{pmatrix} \frac{\partial U}{\partial p} \end{pmatrix}_{T} = 0$ 

U does not change either when p changes at T = C. So the internal energy of an ideal gas is a function of temperature only.

Another important point to note is that for an ideal gas

$$pV = n\overline{R}T$$
 and  $T\left(\frac{\partial p}{\partial T}\right) - p = 0$   
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Therefore

#### $dU = C_v dT$

holds good for an ideal gas in any process (even when the volume changes). But for any other substance

$$dU = C_v d'_z$$

is true only when the volume is constant and dV = 0Similarly

$$dH = TdS + Vdp$$
  
and 
$$TdS = C_{p} dT - T\left(\frac{\partial V}{\partial T}\right)_{p} dp$$
$$\therefore dH = C_{p} dT + \left[V - T\left(\frac{\partial V}{\partial T}\right)_{p}\right] dp$$
$$\therefore \left(\frac{\partial H}{\partial p}\right)_{T} = V - T\left(\frac{\partial V}{\partial T}\right)_{p}$$

As shown for internal energy, it can be similarly proved from Eq. shown in above that the enthalpy of an ideal gas is not a function of either volume or pressure.

but a function of temperature alone Since for an ideal gas,  $pV = n\overline{R}T$ 

the relation of I Cold I is true for any process (even when the pressure changes.)

However, for any other substance tree relation  $dH = C_p dT$  holds good only when the pressure remains constant or dp = 0.

(b) Thermal radiation in equilibrium with the enclosing walls processes an energy that depends only on the volume and temperature. The energy density (u), defined as the ratio of energy to volume, is a function of temperature only, or

$$u = \frac{U}{V} = f(T)only.$$

The electromagnetic theory of radiation states that radiation is equivalent to a photon gas and it exerts a pressure, and that the pressure exerted by the black body radiation in an enclosure is given by

$$p = \frac{u}{3}$$

Black body radiation is thus specified by the pressure, volume and temperature of the radiation. since.

$$U = uV \text{ and } p = \frac{u}{3}$$
$$\left(\frac{\partial U}{\partial V}\right)_{T} = u \text{ and } \left(\frac{\partial p}{\partial T}\right)_{V} = \frac{1}{3}\frac{du}{dT}$$

By substituting in the energy Eq.

$$u = \frac{T}{3}\frac{du}{dT} - \frac{u}{3}$$
$$\therefore \frac{du}{u} = 4\frac{dT}{T}$$

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#### J. Efficiency of Binary vapour cycle:

 $1 - \eta = (1 - \eta_1) (1 - \eta_2) \dots (1 - \eta_n)$ 

 $\therefore$  For two cycles

 $\eta = n_1 + n_2 - n_1 n_2$ 

#### K. Overall efficiency of a power plant

 $\eta_{\rm overall} = \eta_{\rm boiler} \times \eta_{\rm cycle} \times \eta_{\rm turbine \ (mean)} \times \eta_{\rm generator}$ 

# **Questions with Solution P. K. Nag**

#### Q. 12.1 for the following steam cycles find

(a)  $W_T$  in kJ/kg

(b)  $W_p$  in kJ/kg,

(c)  $Q_1$  in kJ/kg,

- (d) cycle efficiency,
- (e) steam rate in kg/kW h, and
- (f) moisture at the end of the turbine process. Show the results in tabular form with your comments.

Boiler Outlet	Condongon Droggy	So Type of Cycle	
	Condenser Pressure		
10 bar, saturated	1 bar	Ideal Ionkine Cycle	
-do-	ldb- 196 C	Neglect $W_p$	
dev	-done	Assume 75% pump and	
	39-	Turbine efficiency	
-do-	0.1 bar	Ideal Rankine Cycle	
10 bar, 300°c	-do-	-do-	
150 bar, 600°c	-do-	-do-	
-do-	-do-	Reheat to $600^{\circ}C$	
		maximum intermediate	
		pressure to limit end	
		moisture to 15%	
-do-	-do-	-do- but with 85% tur- bine	
		efficiencies	
10 bar, saturated	0.1 bar	Isentropic pump process ends	
		on satura	
Boiler Outlet	Condenser Pressure	Type of Cycle	
10 bar, saturated	0.1 bar	-do- but with 80% machine	
		efficiencies	
-do-	-do-	Ideal regenerative cycle	
-do-	-do-	Single open heater at	
		110°c	
-do-	-do-	Two open heaters at 90°	
		and $135^{\circ}c$	
-do-	-do-	-do- but the heaters are	
		closed heaters	
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$$\begin{split} h_{f} &= 191.83 + \left(\frac{15-13.342}{15-10}\right) (225.94 - 191.83) = 203.14 \text{ kJ/kg} \\ h_{fg} &= 2392.8 + \left(\frac{15-13.342}{15-10}\right) (2373.1 - 2392.8) = 2386.3 \text{ kJ/kg} \\ h_{2s} &= h_{f} + x h_{fg} = 203.14 + 0.85033 \times 2386.3 = 2232.3 \text{ kJ/kg} \\ \eta_{isentropic} &= \frac{h_{1} - h_{2'}}{h_{1} - h_{2s}} \\ \therefore \qquad h_{1} - h_{2'} = \eta_{isentropic} \times (h_{1} - h_{2s}) \\ \therefore \qquad h_{2} = h_{1} - \eta_{isentropic} (h_{1} - h_{2s}) \\ = 2803.4 - 0.75 (2803.4 - 2232.3) = 2375 \text{ kJ/kg}. \\ \therefore \qquad \text{Turbine work (W_{T})} = h_{1} - h_{2'} = (2803.4 - 2373) \% \\ &= 428.36 \text{ kJ/kg} \\ \therefore \qquad \text{Efficiency of the plant} = \frac{W_{T}}{h_{1}} = \frac{428.36}{2803.4} \approx 0.1528 = 25.28\% \\ \text{If mass flow rate is } \dot{m} \text{ kg/s} \\ \dot{m} \cdot W_{T} = 12.5 \times 10^{3} \\ \text{or} \qquad \dot{m} = \frac{12.5 \times 10^{3}}{428.86} = 20.42 \text{ kg/s} = 0.1528 = 25.28\% \\ \text{A simple steam power cycle uses solar energy for the heat input} \\ \end{array}$$

Q.12.3

A simple steam power cycle uses solar energy for the heat input. Water in the cycle enters th pump as a saturated liquid at 40°C, and is pumped to 2 bar. It then evaporates in the boiler at this pressure, and enters the turbine as saturated vapour. At the turbine exhaust the conditions are 40°C and 10% moisture. The flow rate is 150 kg/h. Determine (a) the turbine isentropic efficiency, (b) the net work output (c) the cycle efficiency, and (d) the area of solar collector needed if the collectors pick up 0.58 kW/ $m^2$ .

(Ans. (c) 2.78%, (d) 18.2  $m^2$ )



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of 40°C. The heat released by mercury condensing at 0.2 bar is used to impart the latent heat of vaporization to the water in the steam cycle. Mercury enters the mercury turbine as saturated vapour at 10 bar. Compute (a) kg of mercury circulated per kg of water, and (b) the efficiency of the combined cycle.

р	<b>T(°C)</b>	$h_{f}^{}({ m kJ/kg})$	s <sub>f</sub> (kJ/kg k)	$v_f (m^3 / kg)$
(bar)		$h_{_g}$	${m s}_g$	$v_{g}$
10	515.5	72.23	0.1478	<b>80.9</b> x 10 <sup>-6</sup>
		363.0	0.5167	0.0333
0.2	277.3	38.35	0.0967	<b>77.4</b> $\mathbf{x}10^{-6}$
		336.55	0.6385	1.163

The property values of saturated	l mercury are given below
----------------------------------	---------------------------

Solution: Try please.

Solution:

- Q.12.10 In an electric generating station, using a binary vapour cycle with mercury in the upper cycle and steam in the lower, the ratio of mercury flow to steam flow is 10 : 1 on a mass basis. At an evaporition rate of 1,000,000 kg/h for the mercury, its specific enthalpyriset by 356 kJ/kg in passing through the boiler. Superheating the steam of the boiler furnace adds 586 kJ to the steam specific entrative the mercury gives up 251.2 kJ/kg during condensation, and the steam gives up 2003 kJ/kg in its condenser. The overall boller efficiency to 556. The combined turbine metrical and generator efficiencies are each 95% for the mercury and steam only. The steam multiplies require 5% of the energy generated by the units. Find the corrad efficiency of the plant.
- Q.12.11 A sodium-mercury-steam cycle operates between  $1000^{\circ}$ C and  $40^{\circ}$ C. Sodium rejects heat at  $670^{\circ}$ C to mercury. Mercury boils at 24.6 bar and rejects heat at 0.141 bar. Both the sodium and mercury cycles are saturated. Steam is formed at 30 bar and is superheated in the sodium boiler to 350°C. It rejects heat at 0.0 8 bar. Assume isentropic expansions, no heat losses, and no generation and neglect pumping work. Find (a) the amounts of sodium and mercury used per kg of steam, (b) the heat added and rejected in the composite cycle per kg steam, (c) the total work done per kg steam. (d) the efficiency of the composite cycle, (e) the efficiency of the corresponding Carnot cycle, and (f) the work, heat added, and efficiency of a supercritical pressure steam (single fluid) cycle operating at 250 bar and between the same temperature limits. For mercury, at 24.6 bar,  $h_g = 366.78$  kJ/kg

 $s_g = 0.48 kJ/kg K$  and at 0.141 bar,  $s_j = 0.09$ And  $s_g = 0.64 kJ/kg K$ ,  $h_j = 36.01$  and  $h_g = 330.77 kJ/kg$ For sodium, at 1000°C,  $h_g = 4982.53 kJ/kg$ At turbine exhaust = 3914.85 kJ/kg At 670°C,  $h_f = 745.29 kJ/kg$ 

For a supercritical steam cycle, the specific enthalpy and entropy at the turbine inlet may be computed by extrapolation from the steam tables. Try please.

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 $s_f = 1.5301, \ s_{fg} = 5.5967$  $s_{\sigma} = 7.127 \text{ kJ/kg} - \text{K}$  so at point (2) Steam is saturated vapour  $h_2 = 2706.3 \text{ kJ/kg}$ So At 2 bar saturated temperature is 120.2°C but 65°C liquid So  $h_3 = h_2 - C_P \Delta T$  $= 504.7 - 4.187 \times (120.2 - 65) = 273.6 \text{ kJ/kg}$  $W_{p_2} = v_3 (p_1 - p_2) = 0.001 \times (2000 - 200) = 1.8 \text{ kJ/kg}$ :.  $h_4 = 275.4 \text{ kJ/kg}$ :. Heat input (Q) =  $h_1 - h_4 = (3247.6 - 275.4) = 2972.2 \text{ kJ/kg}$ Turbine work =  $(h_1 - h_2) \eta = (3247.6 - 2706.3) \times 0.7 \text{ kJ/kg}$ = 378.9 kJ/kgHeat rejection that utilized (Q<sub>0</sub>) =  $(h_2 - h_3) \eta$  $= (2706.3 - 273.6) \times 0.9 = 2189.4 \text{ kJ/kg}$ ∴ Fraction at energy supplied utiligdale. CO. UK  $\therefore$  Net work output (W<sub>net</sub>) = W<sub>T</sub> - W<sub>P</sub> = 378.9 - 1.8 Preview (b) At 0 59,  $s_{fg} = 7.717$  $\therefore$  Dryness fraction x,  $0.559 + x \times 7.717 = 7.127$ x = 0.85137*.*.. *.*..  $h_2 = 163.4 + 0.85137 \times 2409.1 = 2214.4 \text{ kJ/kg}$  $h_3 = 163.4 \text{ kJ/kg}$  $W_P = 0.001007 \times (2000 - 7) = 2 \text{ kJ/kg}$  $h_4 = 165.4 \text{ kJ/g}.$ *.*..  $W_T = (h_1 - h_2) \times 0.7 = 723.24 \text{ kJ/kg}$ *.*..  $W_{net} = W_T - W_P = 721.24 \text{ kJ/kg}$ Here heat input for power =  $(h_1 - h_4) = 3082.2$ . kJ/kg For same 377.1 kg power we need 0.52285 kg of water So heat input = 1611.5 kJ for power Heat input for heating =  $\frac{2189.4}{0.9}$  kJ = 2432.7 kJ :. Fraction of energy used =  $\frac{377.1 + 2189.4}{1611.5 + 2432.7} \times 100\%$ = 63.46%

Q.12.16 In a nuclear power plant saturated steam at 30 bar enters a h.p. turbine and expands isentropically to a pressure at which its quality is 0.841. At this pressure the steam is passed through a moisture separator which removes all the liquid. Saturated vapour leaves the separator and is Page 206 of 265

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 $h_{\rm f} = 0.559$ ,  $s_{fg} = 7.717$  $\therefore$  If quality is x then  $7.6642 = 0.559 + x \times 7.717 \Rightarrow x = 0.9207$  $\therefore$  h<sub>4</sub> = 163.4 + 0.9207 × 2409.1 = 2381.5 kJ/kg  $h_5 = 163.4 \text{ kJ/kg} \approx h_6$ ,  $h_7 = 686.8 \text{ kJ/kg} \approx h_{\circ}$ : From Heat balance of heater  $m \times h_2 + (1 - m) h_6 = h_7$  $\therefore$  m = 0.2016 kg/kg of steam at H.P  $\therefore (1-m) = 0.7984$  $W_T = h_1 - h_2 + (1 - m) (h_3 - h_4) = 1347.6 kJ/kg$ W<sub>P</sub> neglected  $Q = (h_1 - h_8) + (1 - m) (h_3 - h_2) = 3118.5 \text{ kJ/kg at H.P}$ : (a) Reheat pr. 6.6 bar sale.co.uk (b) Steam flow rate at H.P =  $\frac{80 \times 10^3}{1347.6}$  kg/s = 59.36 kg/s (c) Cycle efficiency  $(\eta) = \frac{W}{Q} = \frac{1347.6}{3118.5}$ 

Q.12.18 Figure shows the air ingement of a stam plant in which steam is also required form industrial nearing process. The steam leaves boiler B at 30 by, 820°C and expands in the H.P. turbine to 2 bar, the efficiency of the H.P. turbine 2 are 75%. At this point one half of the steam passes to the process heater P and the remainder enters separator S which removes all the moisture. The dry steam enters the L.P. turbine at 2 bar and expands to the condenser pressure 0.07 bar, the efficiency of the L.P. turbine being 70%. The drainage from the



Fig. 12.51

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 $\frac{0.025715 \times 0.287 (813 - 543) \ln 2}{4.985} \times 100\% = 27.7\%$ 

Work done = 1.3812 kJ = 53.71 kJ/kg

For ideal regeneration

$$\eta = 1 - \frac{543}{813} = 33.21\%$$

Q13.2 An Ericsson cycle operating with an ideal regenerator works between 1100 K and 288 K. The pressure at the beginning of isothermal compression is 1.013 bar. Determine (a) the compressor and turbine work per kg of air, and (b) the cycle efficiency.

(Ans. (a)  $W_T = 465 \text{ kJ/kg}$ ,  $W_C = 121.8 \text{ kJ/kg}$  (b) 0.738)



 $= 0.7382 \times 1.005 (1100 - 288) \text{ kJ/kg} = 602.4 \text{ kJ/kg}$ 

: 
$$Q_{2-3} = C_P (T_3 - T_2) = C_v (T_3 - T_2) + p_2 (V_3 - V_2)$$

Q13.5 An engine equipped with a cylinder having a bore of 15 cm and a stroke of 45 cm operates on an Otto cycle. If the clearance volume is  $2000 \text{ cm}^3$ , compute the air standard efficiency.

(Ans.47.4%)

## **Chapter 13**

Q13.14 Obtain an expression for the specific work done by an engine working on the Otto cycle in terms of the maximum and minimum

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Temperatures of the cycle, the compression ratio  $\mathcal{K}_{k}$ , and constants of the working fluid (assumed to be an ideal gas).

Hence show that the compression ratio for maximum specific work output is given by

$$r_{L} = \left(\frac{T_{min}}{T_{max}}\right)^{1/2(-\gamma)}$$
Solution:  

$$T_{max} = T_{3}$$

$$Q_{1} = C_{v} (T_{3} - T_{2})$$

$$Q_{2} = C_{v} (T_{4} - T_{1})$$

$$\therefore W = Q_{1} - Q_{2}$$

$$= C_{v} [(T_{3} - T_{2}) - (T_{4} - T_{1})]$$
Hence  $\frac{T_{2}}{T_{1}} = \left(\frac{v_{1}}{v_{2}}\right)^{\gamma-1} = r_{c}^{\gamma-1}$ 
NoteSale.co.uk  
And  $\frac{T_{4}}{T_{3}} = \left(\frac{v_{3}}{v_{1}}\right)^{\gamma-1} = r_{c}^{\gamma-1}$ 
And  $\frac{T_{4}}{T_{3}} = \left(\frac{v_{3}}{v_{1}}\right)^{\gamma-1} = \left(\frac{v_{1}}{v_{2}}\right)^{\gamma-1} = r_{c}^{\gamma-1}$ 
Then  

$$W = C_{v} \left[T_{3} - T_{1}x - \frac{T_{3}}{x} + T_{1}\right]$$
For maximum W,  $\frac{dW}{dx} = 0$ 

$$\therefore C_{v} \left[0 - T_{1} + \frac{T_{3}}{x^{2}} + 0\right] = 0$$

$$\therefore x^{2} = \frac{T_{3}}{T_{1}}$$

$$\therefore r_{c}^{\gamma-1} = \sqrt{\frac{T_{3}}{T_{1}}} = \sqrt{\frac{T_{max}}{T_{max}}}$$

$$\therefore T_{c} = \left(\frac{T_{max}}{T_{min}}\right)^{\frac{1}{2(1-\gamma)}} = \left(\frac{T_{min}}{T_{max}}\right)^{\frac{1}{2(1-\gamma)}}$$
Proved.

Q13.15 A dual combustion cycle operates with a volumetric compression ratio  $r_k$ = 12, and with a cut-off ratio 1.615. The maximum pressure is given by  $p_{max}$  = 54  $p_1$ ' where  $p_1$  is the pressure before compression. Assuming

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indices of compression and expansion of 1.35, show that the m.e.p. of the cycle

$$p_m = 10 p_1$$

Hence evaluate (a) temperatures at cardinal points with  $T_1 = 335$  K, and (b Cycle efficiency.

(Ans. (a) 
$$T_2 = 805$$
 K,  $p_2 = 29.2$   $p_1$  '  $T_3 = 1490$  K,  
 $T_4 = 2410$  K,  $T_5 = 1200$  K, (b)  $\eta = 0.67$ )

Solution:

Here 
$$\frac{v_1}{v_2} = r_c = 12$$
  
 $\frac{v_4}{v_3} = \rho = 1.615$   $pv^{1.35} = C, n = 1.35$   
 $p_{max} = p_3 = p_4 = 54 p_1$ 



$$\begin{array}{ll} \ddots & \frac{T_2}{T_1} = \left(\frac{v_1}{v_2}\right)^{n-1} \therefore T_2 = T_1 \times (12)^{(1.35 - 1)} = 2.3862 \ T_1 \\\\ \text{And} & \frac{p_2}{p_1} = \left(\frac{v_1}{v_2}\right)^n \quad \therefore \ p_2 = \ p_1 \times (12)^{1.35} = 28.635 \ p_1 \\\\ & \frac{p_2}{T_2} = \frac{p_3}{T_3} \qquad \therefore \ T_3 = \ \frac{p_3}{T_2} \times T_2 = \frac{54p_1}{28.635p_1} \times 2.3862 \ T_1 = 4.5 \ T_1 \\\\ v_3 = v_2 = \left(\frac{v_1}{12}\right) \\\\ \ddots & v_4 = \rho \ v_3 = \frac{1.615}{12} \ v_1 = 0.13458 \ v_1 \\\\ \therefore & \frac{p_4 v_4}{T_4} = \frac{P_3 v_3}{T_3} \qquad p_3 = \ p_4 \\\\ & T_4 = \ T_3 \times \frac{v_4}{v_3} = 1.615 \ T_3 = 1.615 \times 4.5 \ T_1 = 7.2675 \ T_1 \\\\ \therefore & \frac{T_5}{T_4} = \left(\frac{v_4}{v_5}\right)^{n-1} = \left(\frac{v_4}{v_1}\right)^{n-1} \\\\ \therefore & T_5 = 3.6019 \ T_1 \\\\ \therefore & W = [C_v (T_3 - T_2) + C_P (T_4 - T_3) - C_v (T_5 - T_1)] = 2.4308 \ T_1 \ kJ/kg. \end{array}$$

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(b) Turbine work  $(W_T) = (h_1 - h_2) = C_p(T_1 - T_2) = 351.75 \text{ kJ/kg}$ 

(c) Heat supplied 
$$(Q_1) = C_p(T_1 - T_4) = 517.6 \text{ kJ/kg}$$

(d) Cycle efficiency (
$$\eta$$
) =  $\frac{W_T - W_C}{Q_1}$  = 17.86%

- (e) Turbine exhaust temperature  $(T_2) = 723$  K
- Q13.27 A simple gas turbine plant operating on the Brayton cycle has air inlet temperature 27°C, pressure ratio 9, and maximum cycle temperature 727°C. What will be the improvement in cycle efficiency and output if the turbine process Is divided into two stages each of pressure ratio 3, with intermediate reheating to 727°C?

Solution:

(Ans. - 18.3%, 30.6%)



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$$\therefore \eta = \frac{99.18}{583.5} \times 100\% = 17\%$$

Q13.29 A gas turbine plant draws in air at 1.013 bar, 10°C and has a pressure ratio of 5.5. The maximum temperature in the cycle is limited to 750°C. Compression is conducted in an uncooled rotary compressor having an isentropic efficiency of 82%, and expansion takes place in a turbine with an isentropic efficiency of 85%. A heat exchanger with an efficiency of 70% is fitted between the compressor outlet and combustion chamber. For an air flow of 40 kg/s, find (a) the overall cycle efficiency, (b) the turbine output, and (c) the air-fuel ratio if the calorific value of the fuel used is 45.22 MJ/kg.

(Ans. (a) 30.4%, (b) 4272 kW, (c) 115)





 $T_c$  = temperature of the cold reservoir.



Compressor power =  $\dot{m}W$  = 7.95 KW

- Q14.26 A 100 tonne low temperature R-12 system is to operate on a 2-stage vapour compression refrigeration cycle with a flash chamber, with the refrigerant evaporating at – 40°C, an intermediate pressure of 2.1912 bar, and condensation at 30°C. Saturated vapour enters both the compressors and saturated liquid enters each expansion valve. Consider both stages of compression to be isentropic. Determine:
  - (a) The flow rate of refrigerant handled by each compressor
  - (b) The total power required to drive the compressor
  - (c) The piston displacement of each compressor, if the clearance is 2.5% for each machine
  - (d) The COP of the system
  - (e) What would have been the refrigerant flow rate, the total work of compression, the piston displacement in each compressor and the compressor and the COP, if the compression had occurred in a single stage? .

(Ans. (a) 2.464, 3.387 kg/s, (b) 123 kW, (c) 0.6274, 0.314 m<sup>3</sup>/s, (d) 2.86, (e) 3.349 kg/s, 144.54 kW, 1.0236 m<sup>3</sup>/s, 2.433)