## VLOF: Lift off speed

- Short runways higher flaps to take-off faster lower VLOF but lowers climb performance
- Distant obstacles lower flaps higher VLOF, higher ground roll distance
- Higher with increased mass

## V<sub>MCG</sub>:

- Min control speed on the ground
- No nose-wheel steering & no crosswind used for determination
- Determined by engine thrust & rudder deflection
- Determined by primary aerodynamic control only
- $V_{MCG} < V_{EF} < V_1$

 $V_1$ :

- Pilot decides to abort take-off AT V<sub>1</sub> (At the last resort)
- Limited by  $V_{MCG}$   $V_{R}$  &  $V_{MBE}$
- Min value  $V_{MCG}$ , max value  $V_{R}$
- Must not be exceeded by V<sub>MBE</sub>
- **Can** be **higher** than  $V_{MU}$
- Value exceeds correct  $V_1$  value = ASD will exceed the ASDA
- $V_1$  increase but  $V_R$  the same = Increased ASD
- Higher value used with constant mass: TODR decrease & ASDR increase
- Reduced by inoperative anti-skid (Because you are braking manually you need a lower decision speed)
- OEI obstacle clearance reduces because of contaminated runway, but climb performance remains constant
- Increased with mass (Because higher mass requires more lift = more speed)
- Down slope decreases V<sub>1</sub>

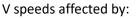
 $V_R$ :

- Speed at which pilot should start to rotate the aeroplane
- If aircraft rotates earlier: Stabilizer trim setting miscalculated, centre of gravity too far aft the calculated V<sub>R</sub> does **not cause** early rotation, it is just a calculated value) Speed to which rotation to the lift off angle is initiated Must not be less than 1.05V<sub>MCA</sub> or V<sub>1</sub> econds are for **recognition** lax brake energy): Must not be exceeded by Val

- V<sub>EF</sub>: 2 seconds are for recognition

V<sub>MBE</sub> (Max brake energy):

- Must not be exceeded by V
- energy thus allows an increased mass (A good thing) If TOM is V<sub>MBE</sub> limited TO U **shill** requi  $V_2$ :
  - Take-of safety speed/take off climb speed or speed at 35ft
  - May not be less than 1.13 V<sub>sR</sub> for turbojets
  - May not be less than 1.08 V<sub>sR</sub> for turboprops
  - May not be less than 1.10 V<sub>MCA</sub>
  - Limited by V<sub>MCA</sub>: Large flap angles, high air pressure & low aircraft weight (What is good for thrust also increases adverse yaw OEI)
  - **Decreases with higher flaps**
  - Increased V<sub>2</sub> procedure(Improved take-off climb/climb performance procedure):
    - Only possible when an excess field length is available (ASD is not limiting)
    - Further screen height along runway •
    - Increases TODR & climb gradient for a given TOM •
  - $V_{2MIN}$ :
    - Decrease with higher flaps if not limited by  $V_{MCA}$
    - Uses V<sub>SR</sub> & V<sub>MCA</sub>



- Mass: Lower mass, lower speeds
- Density altitude: Density altitude increase, thrust decrease, lower  $V_{MCA}$  to counteract yaw OEI
- Low field elevation = lower speeds
- Flap settings: Higher flaps = decreased stall speed

